# Fatigue Crack Growth on Several Materials under Single-Spike Overloads and Aircraft Spectra

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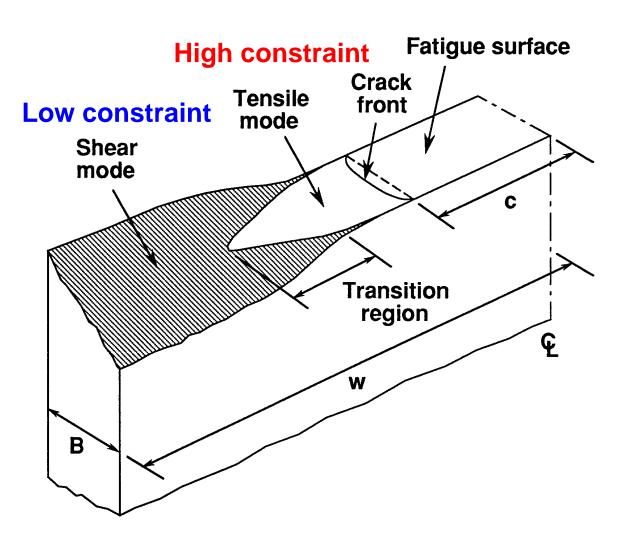
Research Sponsors: Mississippi State University QinetiQ Australia

• Plane-strain to plane-stress fatigue-crack growth behavior

### Flat-to-Slant Crack Growth and the Associated Constraint-Loss Behavior

**Schijve (1966)** ASTM STP-415:

Crack-growth rate was "constant" at transition on 2024-T3 Alclad for wide range in R

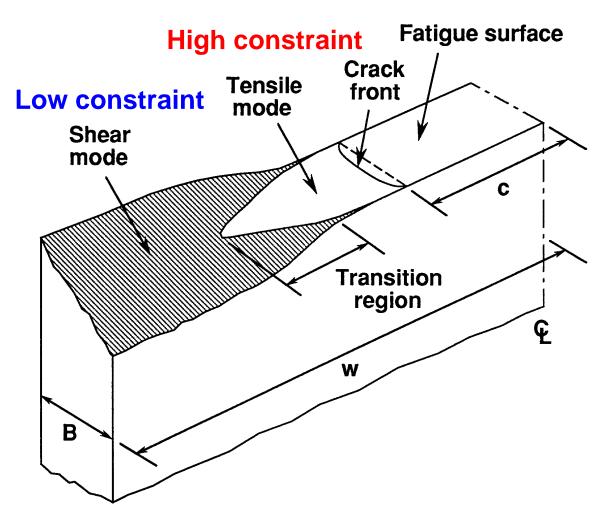


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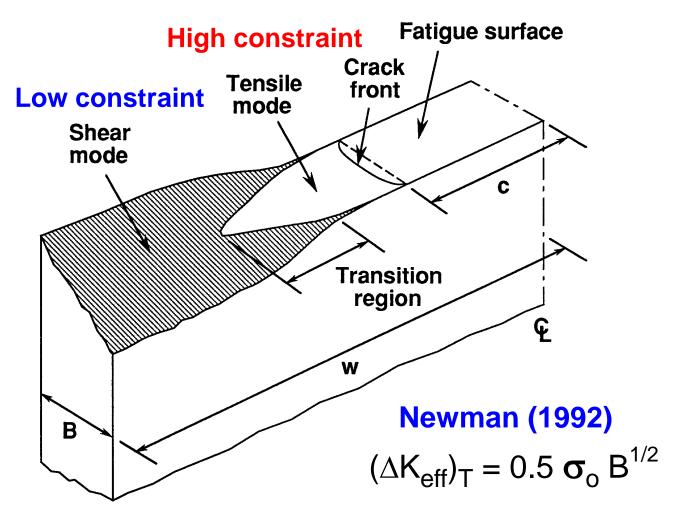


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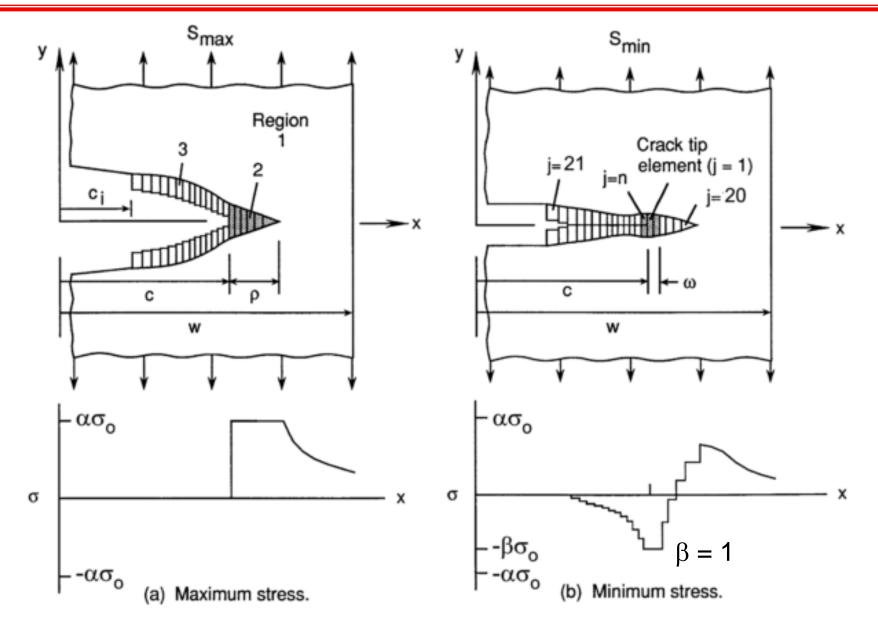
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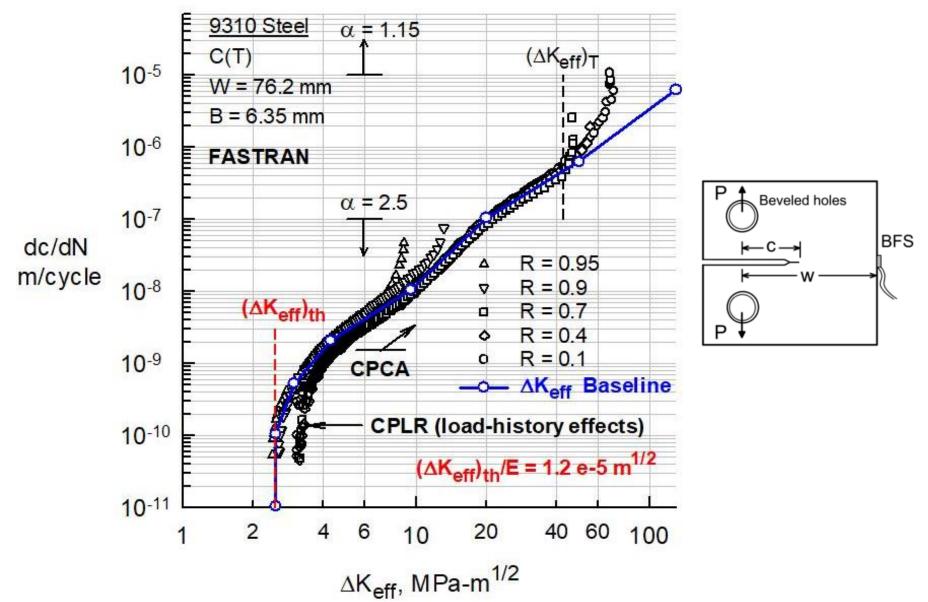


- Plane-strain to plane-stress fatigue-crack growth behavior
- Fatigue-crack-growth-rate against  $\Delta K_{eff}$  correlations

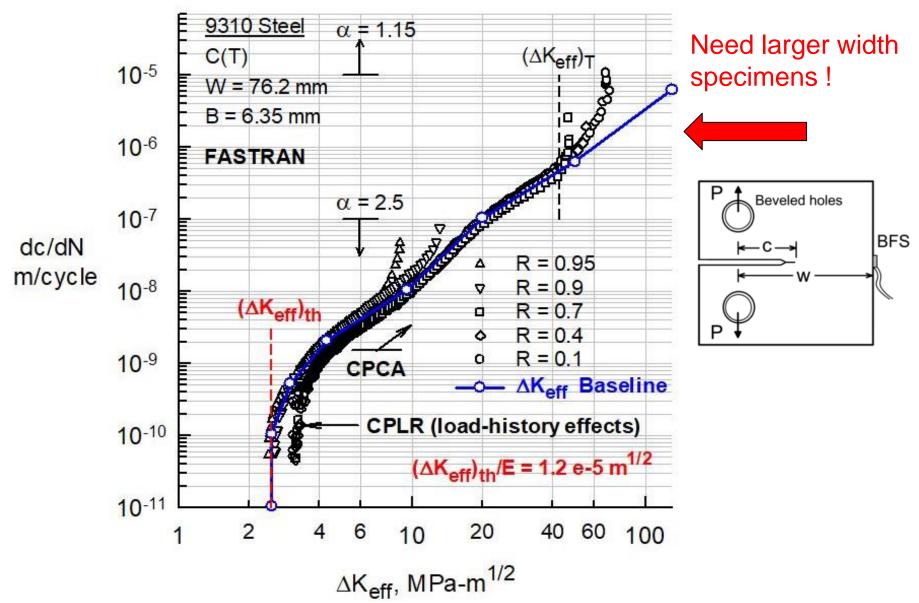
### FASTRAN – Fatigue-Crack-Closure based Life-Prediction Code 1976 - Present



### Effective SIF Relation for 9310 Steel Plate C(T) Specimens

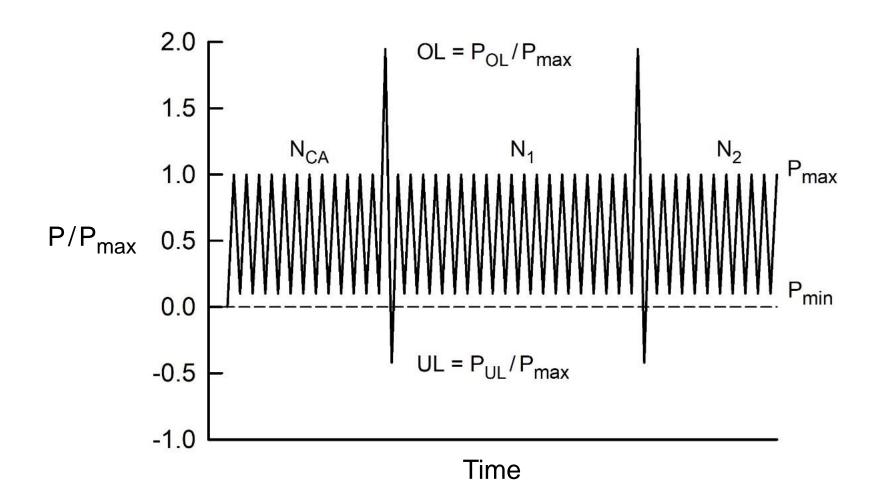


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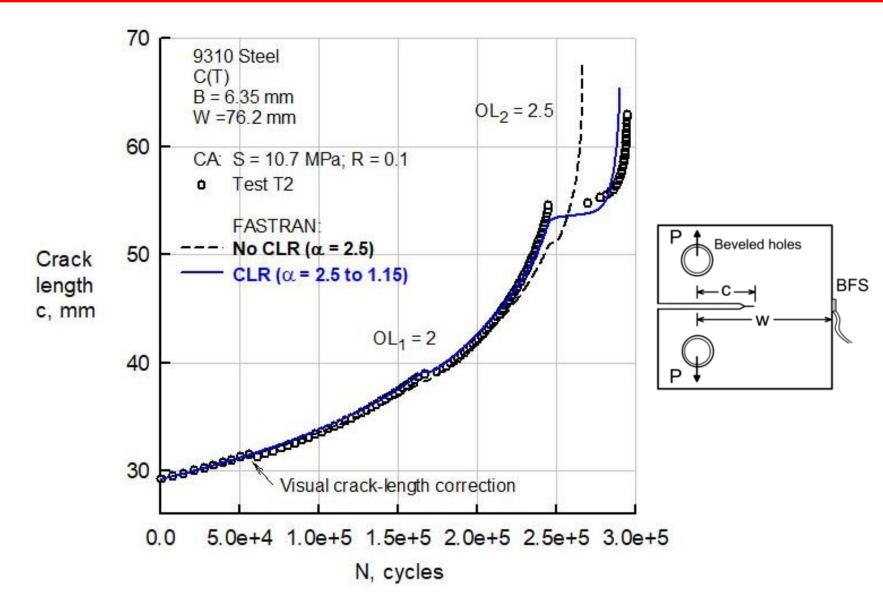


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- Single-spike overload/underload tests and analyses
  - 9310 Steel Plate C(T) Newman et al. (2013)

#### Repeated Single-Spike Overload/Underload History under Constant-Amplitude Loading

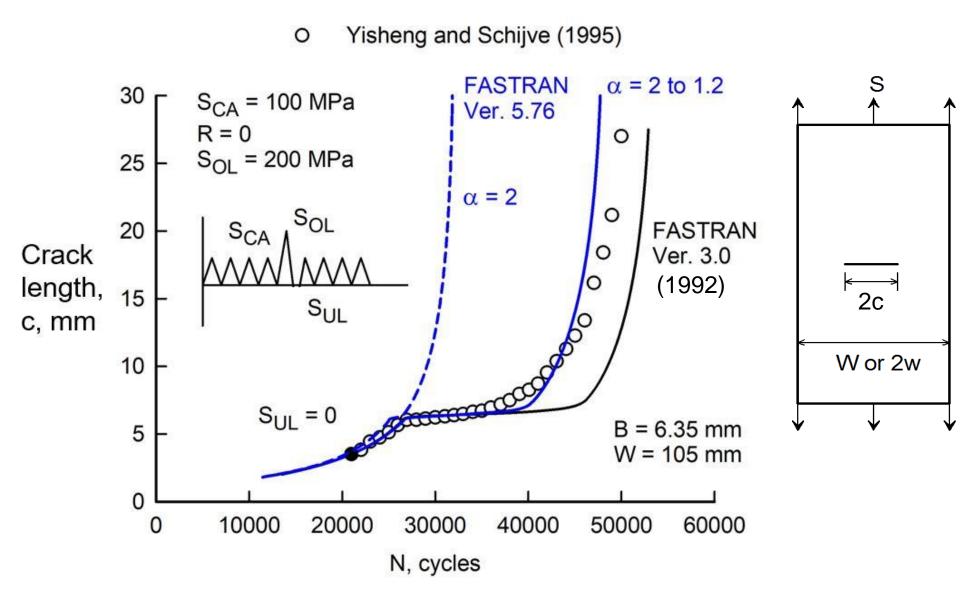


#### Measured and Predicted Crack-Length-against-Cycles for C(T) Specimen made of 9310 Steel Plate under Repeated Single-Spike Overloads



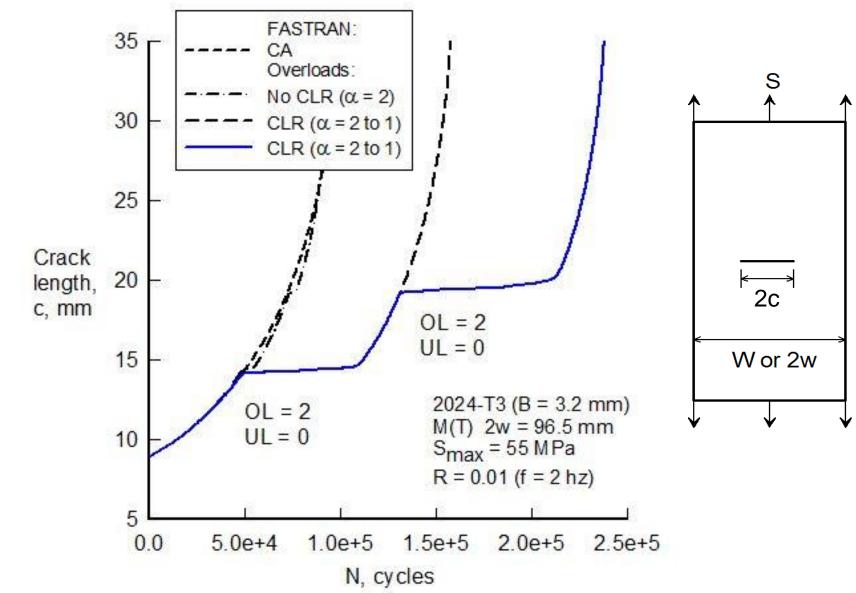
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#### Test and Analyses of a Single-Spike Overload on 2024-T3 Plate

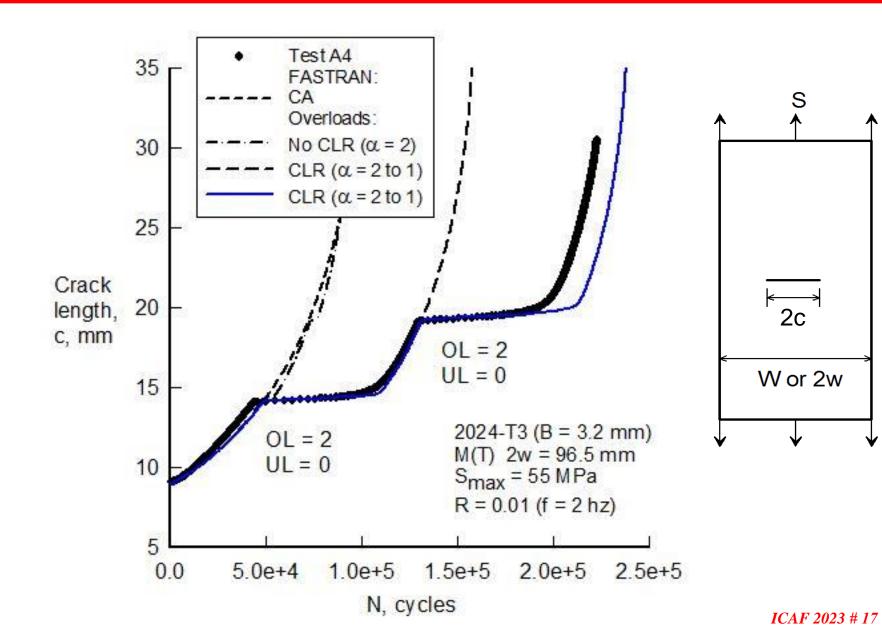


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  - 2024-T3 Sheet M(T) Newman-Walker

#### Predicted Crack-Length against Cycles under Repeated Single-Spike Overloads in 2024-T3 Sheet

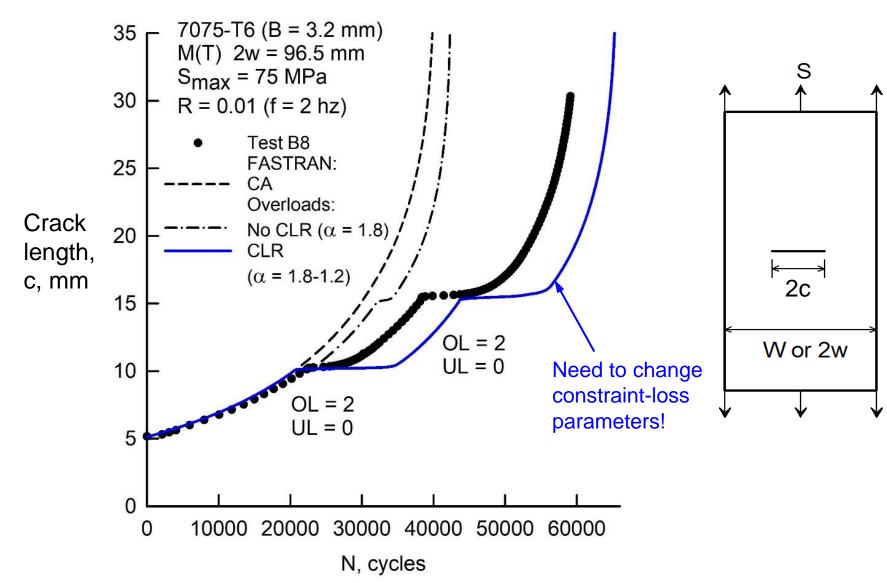


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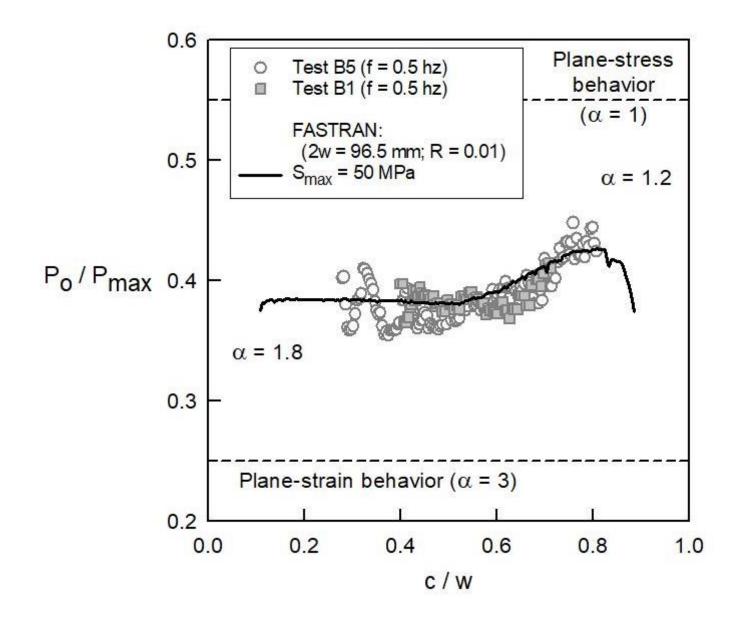


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  - 7075-T6 Sheet M(T) Newman-Walker

#### Measured and Predicted Crack-Length against Cycles under Repeated Single-Spike Overloads in 7075-T6 Sheet



#### Measured and Calculated Constraint-Loss Behavior for 7075-T6 Sheet under Constant-Amplitude Loading

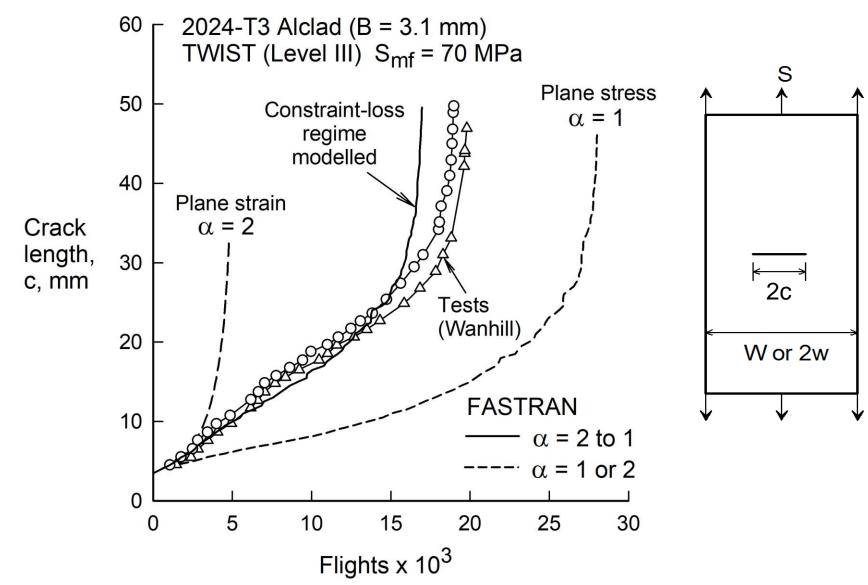


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  - 2024-T3 Sheet M(T) Newman-Walker
  - 7075-T6 Sheet M(T) Newman-Walker

### TWIST spectrum crack-growth tests (Wanhill) and analyses

#### Crack Growth under TWIST (Level III) Spectrum Loading

Wanhill (1977), Newman (1992)



 Transition from plane-strain to plane-stress behavior (flat-to-slant crack growth) occurs at a constant crack-growth rate and is controlled by (∆K<sub>eff</sub>)<sub>T</sub>.

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- **Constraint-loss behavior** from plane-strain to plane-stress behavior was *able* to calculate or predict delays caused by single-spike overloads and underloads.

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- Constraint-loss behavior from plane-strain to plane-stress behavior was *able* to predict crack growth under the TWIST spectrum loading.

Thank You, Very Much !

Questions ?





