ICAF 2023 – the 38th Conference and 31st Symposium of the International Committee on Aeronautical Fatigue and Structural Integrity

Retardation of Fatigue Cracks in Welded Structures through Laser Shock Peening

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OUTLINE

- 1. Introduction and motivation
- 2. Laser processed Al-alloys for damage tolerant design
- 3. Laser processed Ti-alloys for defect tolerant design
- 4. Summary and outlook



Introduction and motivation

Environmental challenges and future technologies

- The goals defined by Flightpath 2050 (compared to 2000 levels)¹:
 - 75% reduction in CO₂ emissions
 - 90% reduction in NO_x emissions
 - 65% reduction in perceived noise

New design concepts Processing technologies **Advanced materials** Metals: propulsion Laser-based techniques for Al-, Ti-, Mg-, TiAl-alloys, joining, additive high-entropy alloys. Composites and hybrid structures. ©Clean Sky airframe © www.thefifthestate.com ©Leonardo

manufacturing and residual stress engineering.

75% of the world fleet

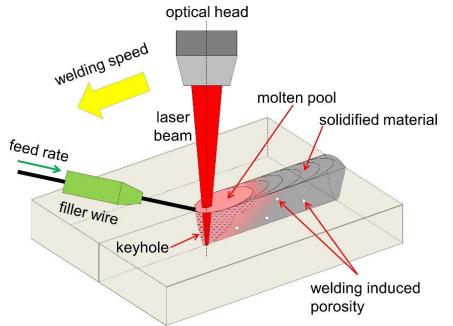
will be replaced by

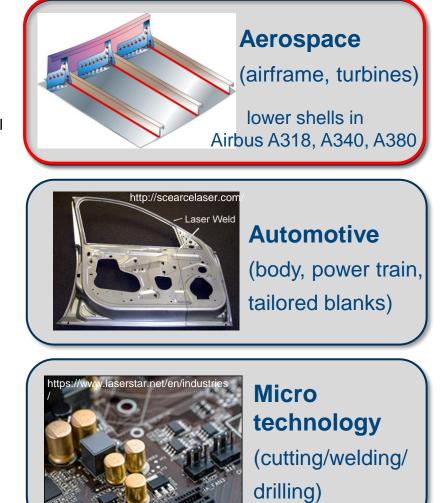
2050



Introduction and motivation

Laser beam welding (LBW). Overview



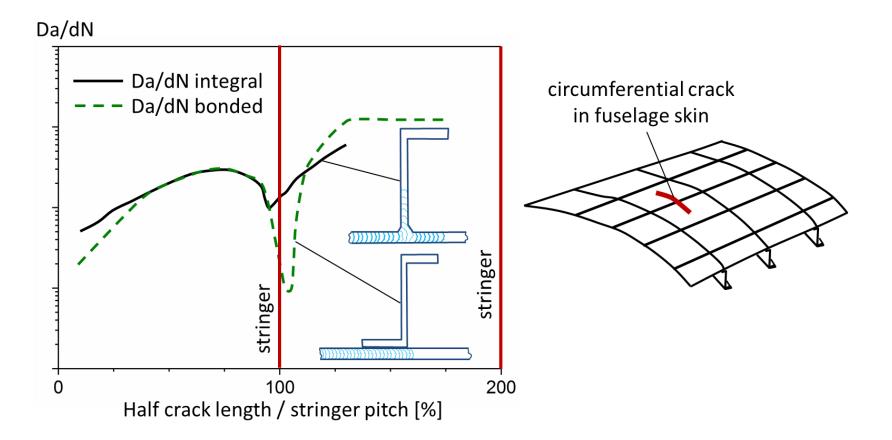




- > single step process \rightarrow high productivity
 - high flexibility of the process
 - > low heat input \rightarrow low distortion
 - > no need for high vacuum
 - > non-contact technique

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Crack scenarios in integral structures and built-up structures

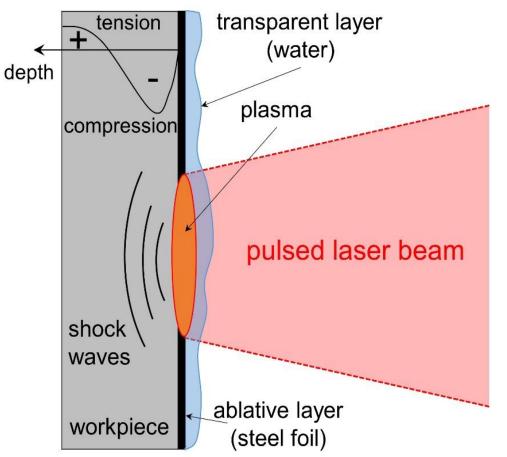


Inferior damage tolerance behaviour of laser welded structures

H.-J. Schmidt & B. Schmidt-Brandecker, ECF 2010.



Laser shock peening for life extension and repair

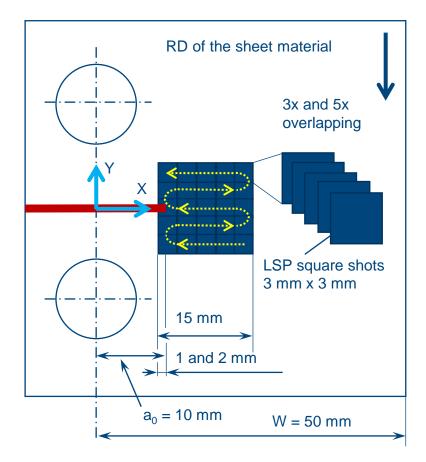


- Tailored residual stress fields
- Low surface roughness
- Retardation or suppression of crack initiation
- Deceleration of crack growth





Experimental set-up for LSP at Hereon



C(T)50, AA2024 T3, thickness = 2 mm

N. Kashaev et al.: Int. J. Fatigue 98 (2017) 223-233

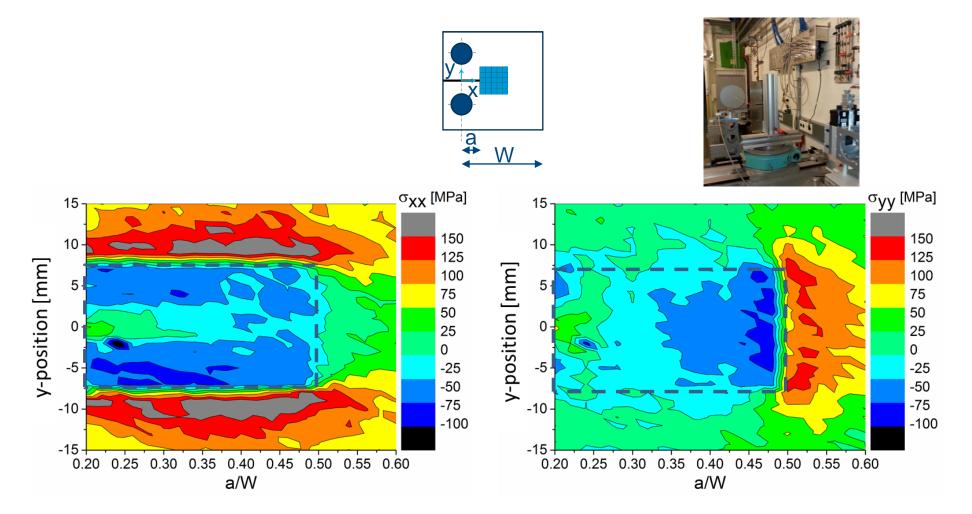


Q-switched Nd:YAG-laser, 3 J, 10 ns, 3 mm x 3 mm square spot, Al-foil **3.3 GW/cm²**

Specimen identification	Numbers of LSP layers	Distance of LSP patterns before the crack tip
BM	-	-
LSP 5x -1	5	1 mm
LSP 3x -2	3	2 mm
LSP 5x -2	5	2 mm



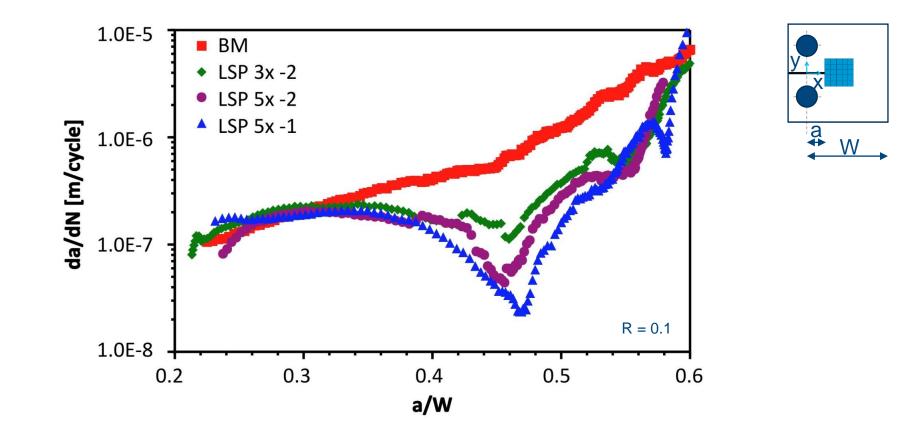
Residual stress analysis using synchrotron radiation



Generation of through-the-thickness compressive residual stresses is possible by using the LSP treatment!

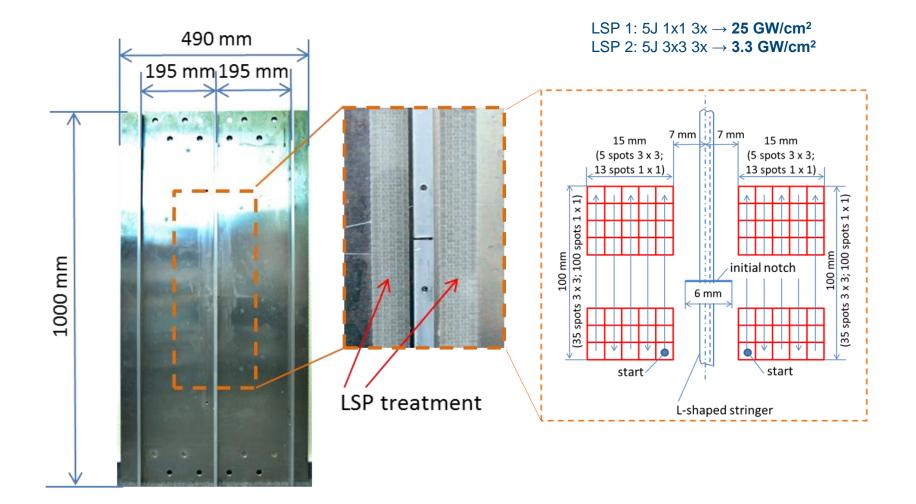


Fatigue crack propagation



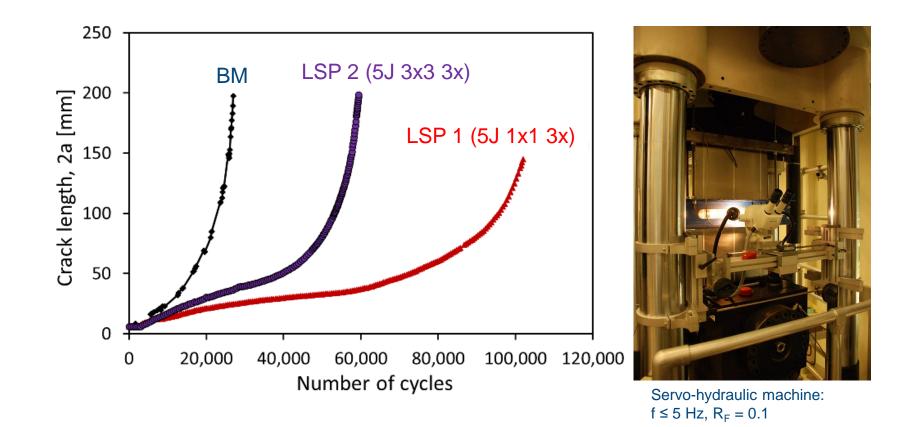
Significant retardation in the FCP when the crack propagates through the LSPtreated area with high compressive through-the-thickness residual stresses!

LSP-treatment of welded AA2024-AA7050 3-stringer panels





FCP-test results of welded 3-stringer AA2024-AA7050 panels

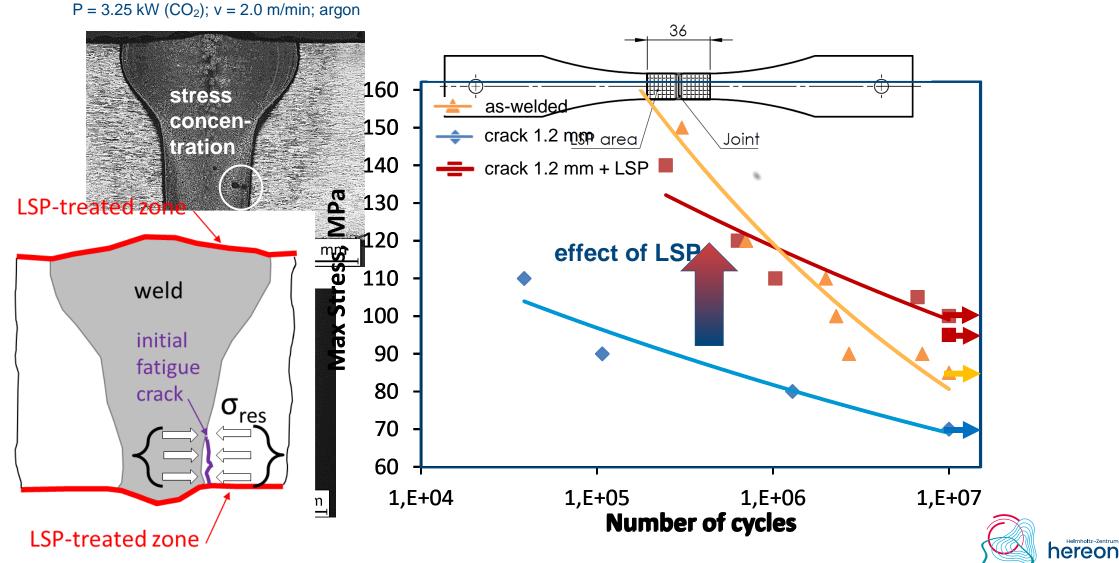


An increase of up to 200-400% in fatigue life could be obtained by local treatment via laser shock peening

N. Kashaev et al.: MATEC Web. Conf. 165 (2018) 18001

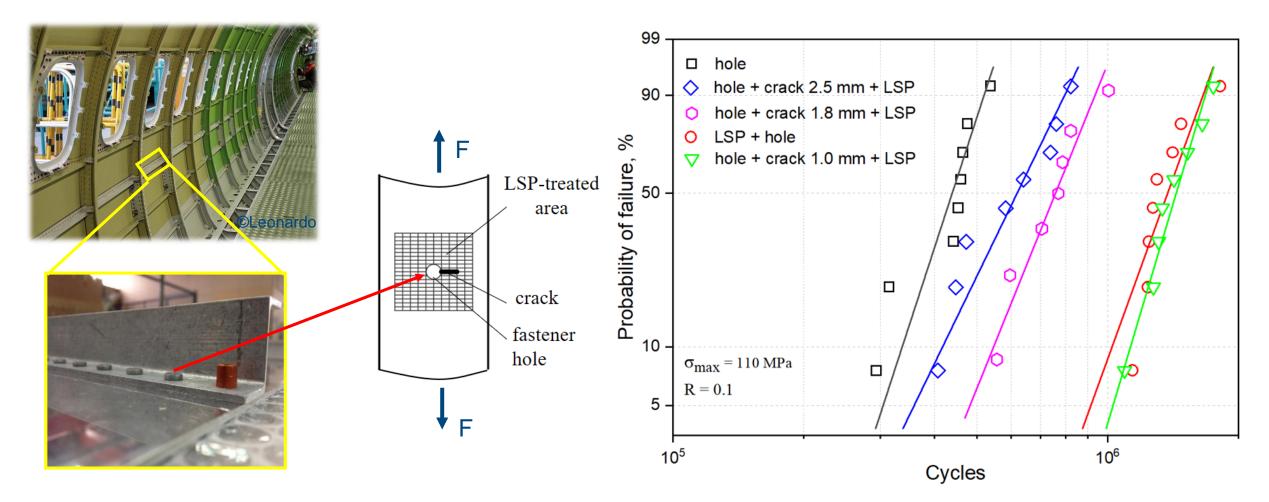


LSP for welds repair – application for laser welded AA6056 with surface fatigue cracks



N. Kashaev et al.: Fatigue. Fract. Eng. Mater. Struct. 2020

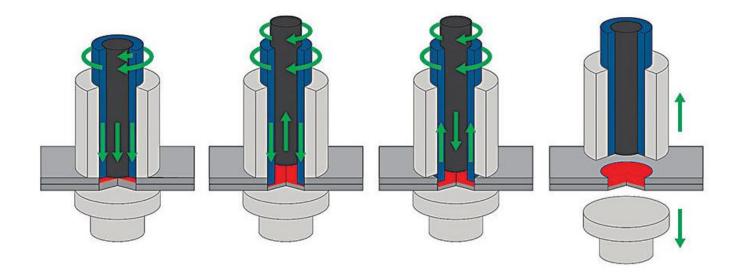
LSP for repair – Application for riveted joints



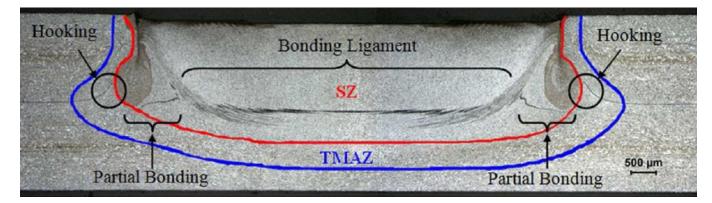
Fatigue life increase of pre-cracked and LSP-treated specimens with a rivet hole by factor 1.5 - 3.3 compared to untreated crack-free specimens



Fatigue Life Extension of Refill Friction Stir Spot-Welded Joints





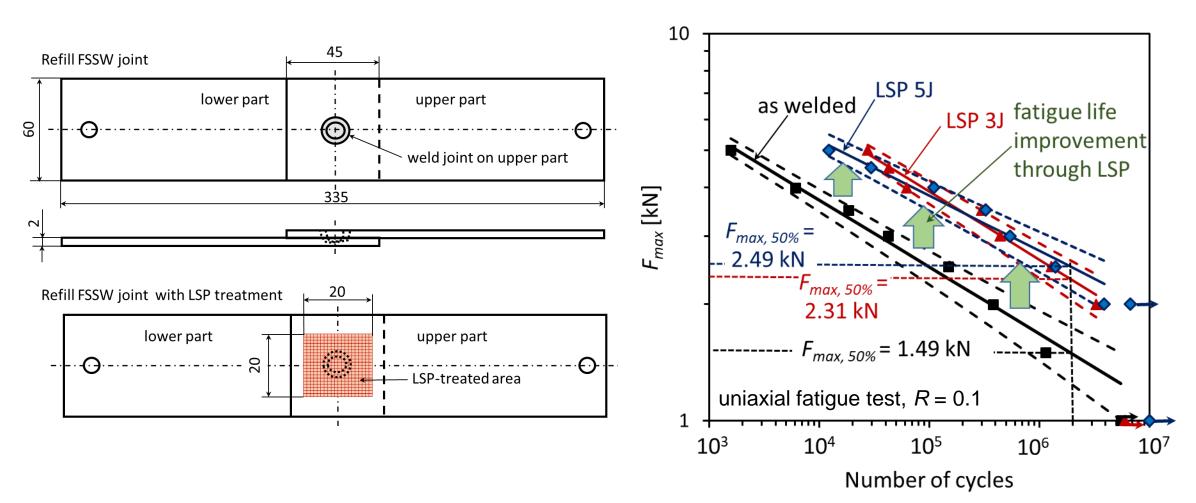


fatigue strength of the refill FSSW joints is only 15% of the ultimate lap shear strength



R. C. Brzostek et al. Fatigue Fract. Eng. Mater. Struct. 41 (2018) 1208-1223

Fatigue Life Extension of Refill Friction Stir Spot-Welded Joints



LSP represents a promising post-processing technique to increase the service life of refill FSSW joints without adding weight to the structure, thus making this innovative joining technique a competitive replacement for classical riveting

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Summary and Outlook

Retardation of Fatigue Cracks in Welded Structures through Laser Shock Peening

 \blacktriangleright An increase of up to 200-400% in fatigue life could be obtained by local treatment via laser shock peening

> Laser shock peening can be applied as an repair and/or a prophylactic residual stress engineering approach to extend the fatigue life of critical structures in ageing aircrafts

> The positive effect of laser shock peening is also demonstrated for riveted and refill FSSWed joints

